

Use of Hydrocarbon-based Alternative Fuels in Gas Turbine Applications

**P. Gokulakrishnan, C. C. Fuller & M. S. Klassen
Combustion Science & Engineering, Inc.**

USAF SBIR Contract # FA8650-09-C-2009
USAF SBIR Contract #

ISSI Subcontract # SB05010 in support of
Prime Air Force Contract # F33615-03-D-2329 DO 0002

**14th ISRAELI SYMPOSIUM ON JET ENGINES & GAS TURBINES
NOVEMBER 5 2015**



Background and Motivation

- Alternatives to petroleum-based fuels are of interest to the aviation and power generation industry , adding another layer of complexity from a combustion behavior and fuel modeling standpoint.
- Alternative jet fuels need to be “drop-in” ready – i.e. function without modifications to the engine components while not affecting engine performance (emissions, stability etc.)
- Alternative fuels for power generation should not affect pollutant emissions (NO_x, CO, particulates) while not greatly affecting engine maintenance schedules
- Examples of alternative jet fuels include:
 - Hydrotreated Renewable Jet Fuel (HRJ)
 - Fischer-Tropsch Fuels (F-T Fuels)
 - E.g. SPK, IPK, S-8

Background and Motivation

- A more complete method to test the combustion and ignition of alternative fuels is to examine the behavior of the fuels across a range of oxidizer conditions.
- Jet fuel injected into turbine exhaust of aircraft engines results in combustion at vitiated conditions which represent more complex combustion regimes compared to standard air.
- Exhaust gas recirculation (EGR) is being examined as a way to control emissions in power generation applications
- Vitiated air in this study is defined as:
 - oxidizer stream with reduced O_2 concentrations typically due to EGR or operation as secondary combustion devices
 - O_2 concentrations as low as 12 – 18 vol% as well as the presence of combustion products including H_2O , CO_2 , CO , and NO_x

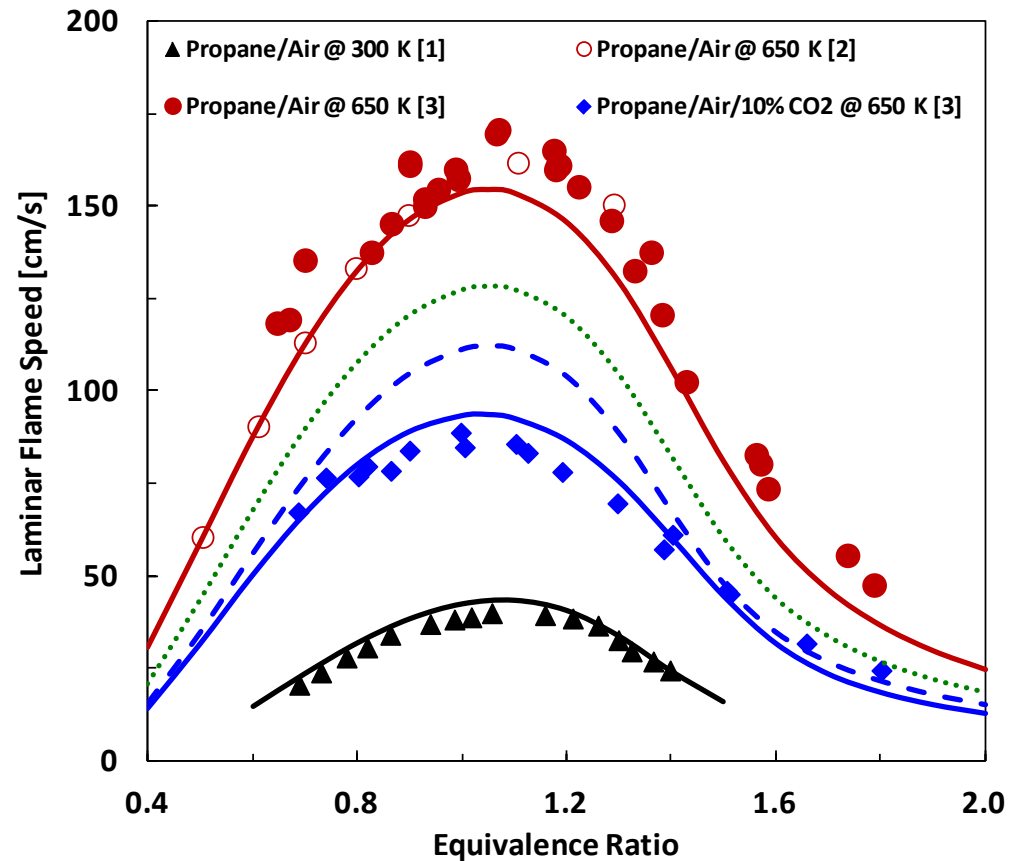
Background and Motivation

- Experimental work has shown that vitiation causes significant variations to combustion processes including
 - flame propagation
 - ignition
 - extinction
 - emissions
- There are limited experimental data available for combustion of liquid hydrocarbon fuels under vitiated conditions
- The focus of this effort is on the ignition behavior of petroleum based JP-8 and its alternatives at low-pressure vitiated conditions.
- The data obtained from this effort will be used to further develop kinetic models for surrogate fuels under normal air and vitiated combustion conditions.

Effect of Vitiation on Flame Speed

- Vitiation has been shown to have significant effects on flame propagation.
- Effects have been found to be based on the thermodynamics and the kinetics of the diluent species present in vitiated air.
- In the case of CO_2 , its presence in propane/air flames reduced flame speed to a greater degree than would be expected based solely on its thermodynamic properties.

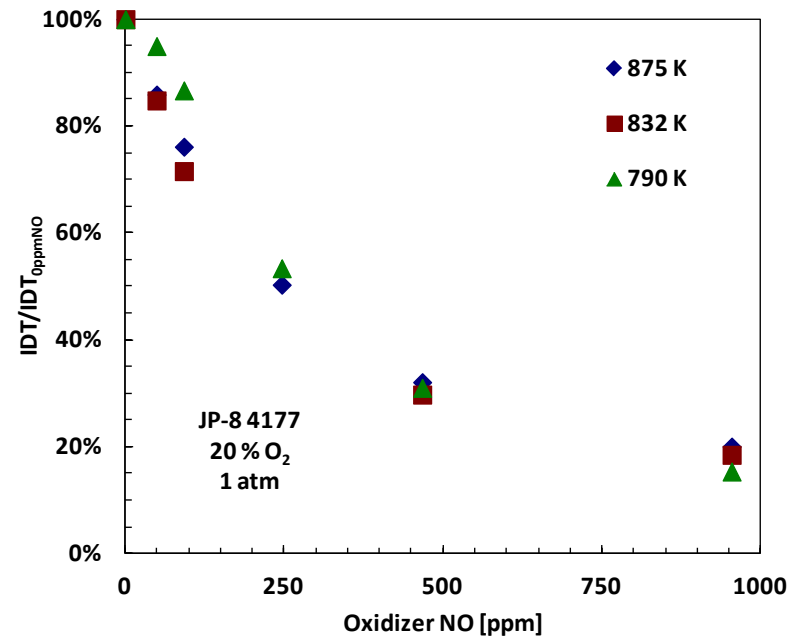
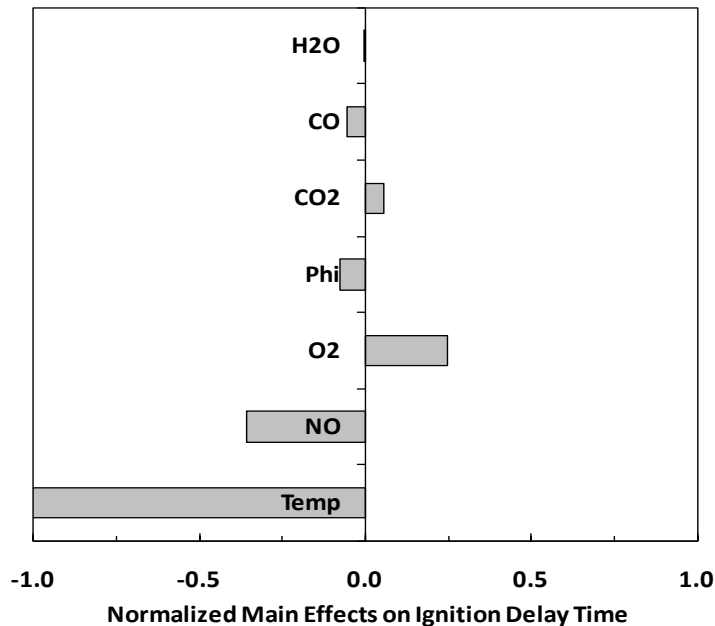
Figure adapted from Gokulakrishnan et al. 2014



Solid Lines: Full Kinetic Model
Dashed Line: Kinetically inert CO_2 added to vitiated air
Dotted Line: 10% N_2 dilution

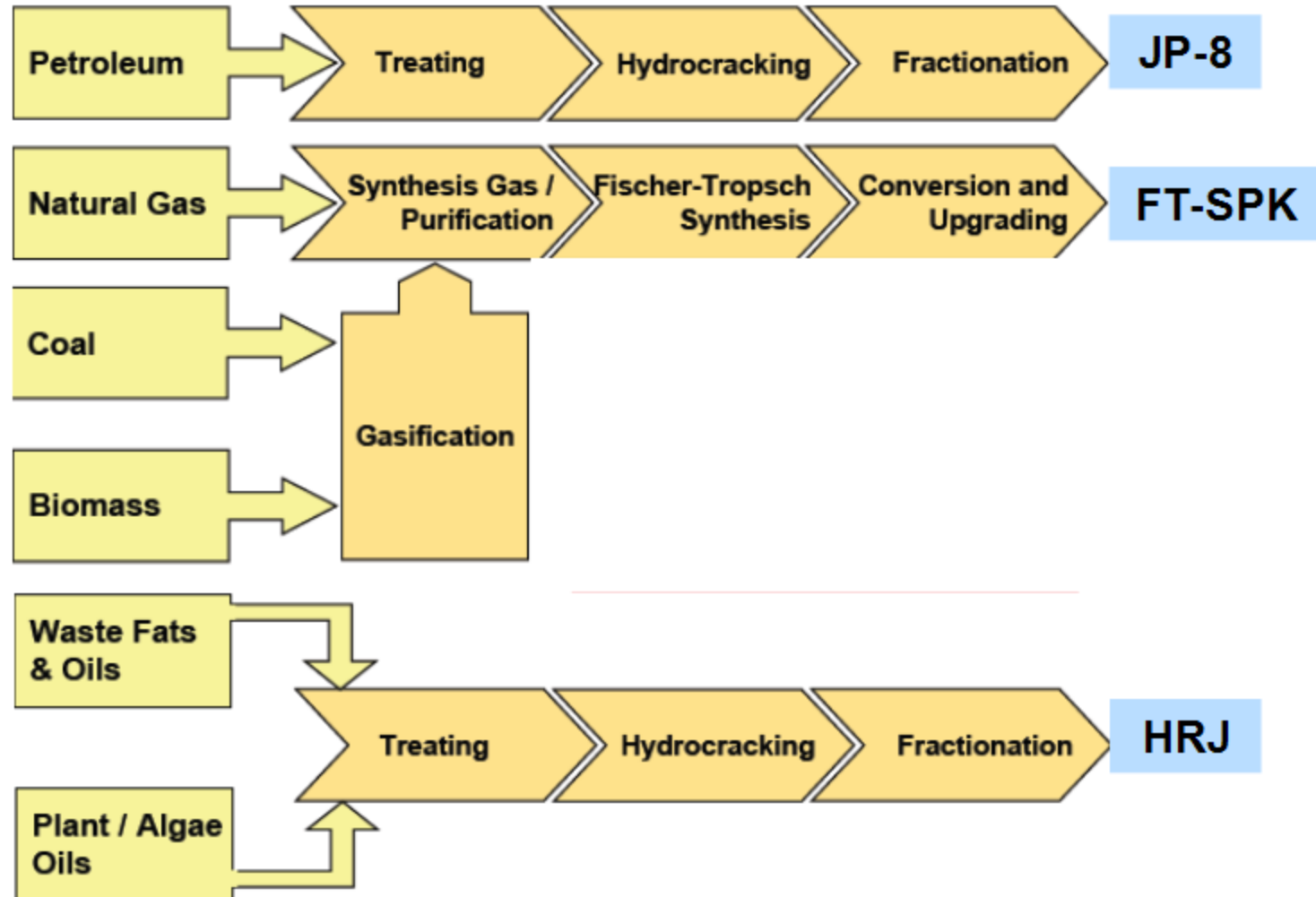
[1] Zhao et al. (2004); [2] Vagelopoulos and Egolfopoulos (1998); [3] Fuller et al. (2012)

Effect of Vitiation on Ignition

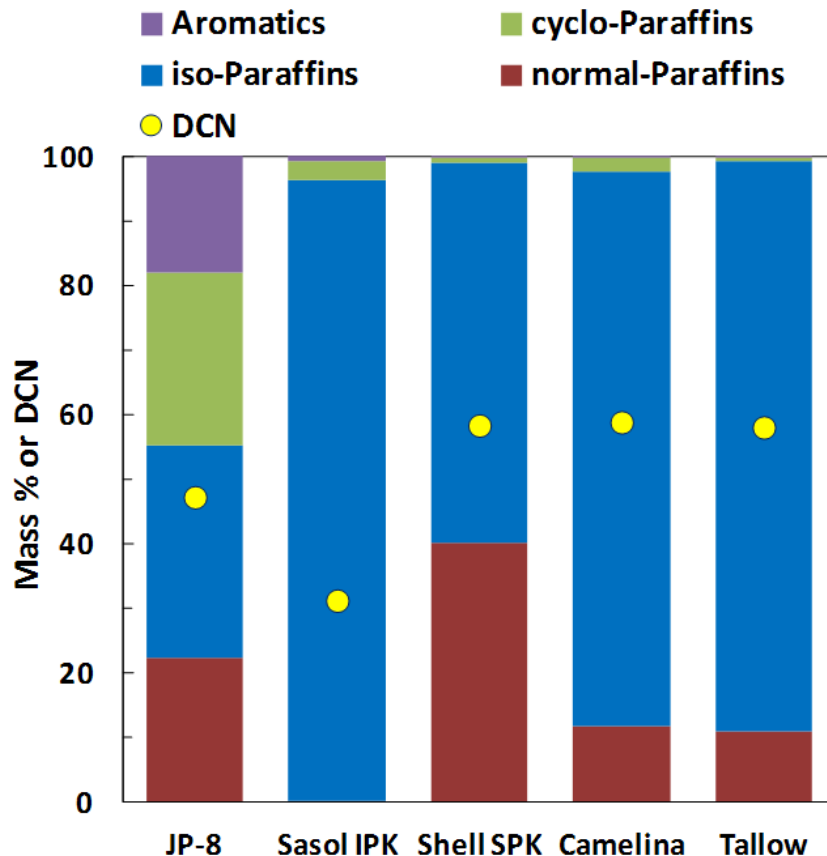


- Previous effort [Fuller et al. 2011] examined the effects of vitiation on ignition delay time of JP-8
- Temperature, O₂ and NO_x were found to have significant effect on the induction chemistry of jet fuels.
- Presence of NO_x specifically reduces ignition delay time quite significantly when it is introduced to the oxidizer stream in relatively small quantities (50 - 1000 ppm)

Alternative Jet Fuels: FT-SPK and HRJ



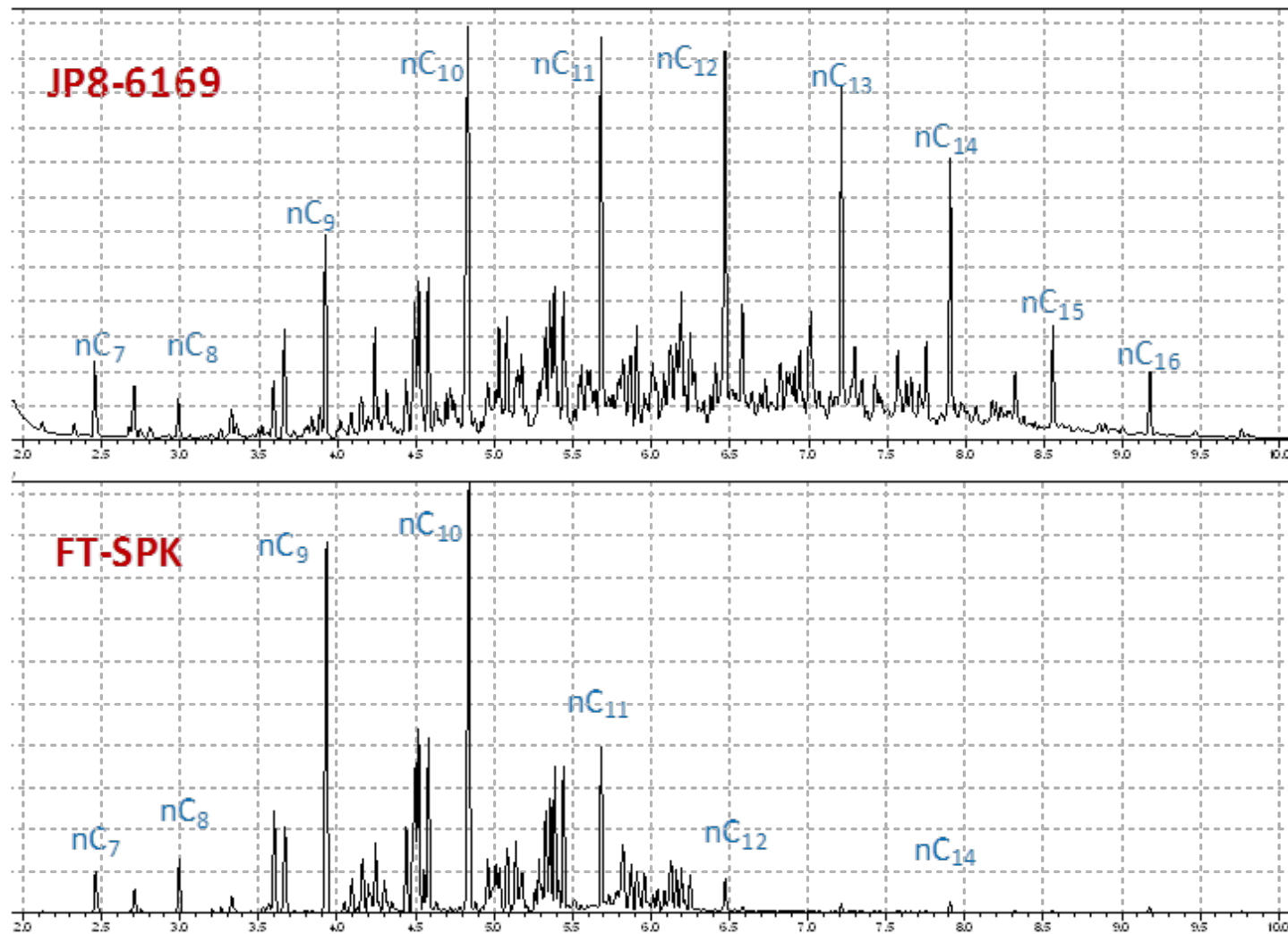
Alternative Jet Fuels



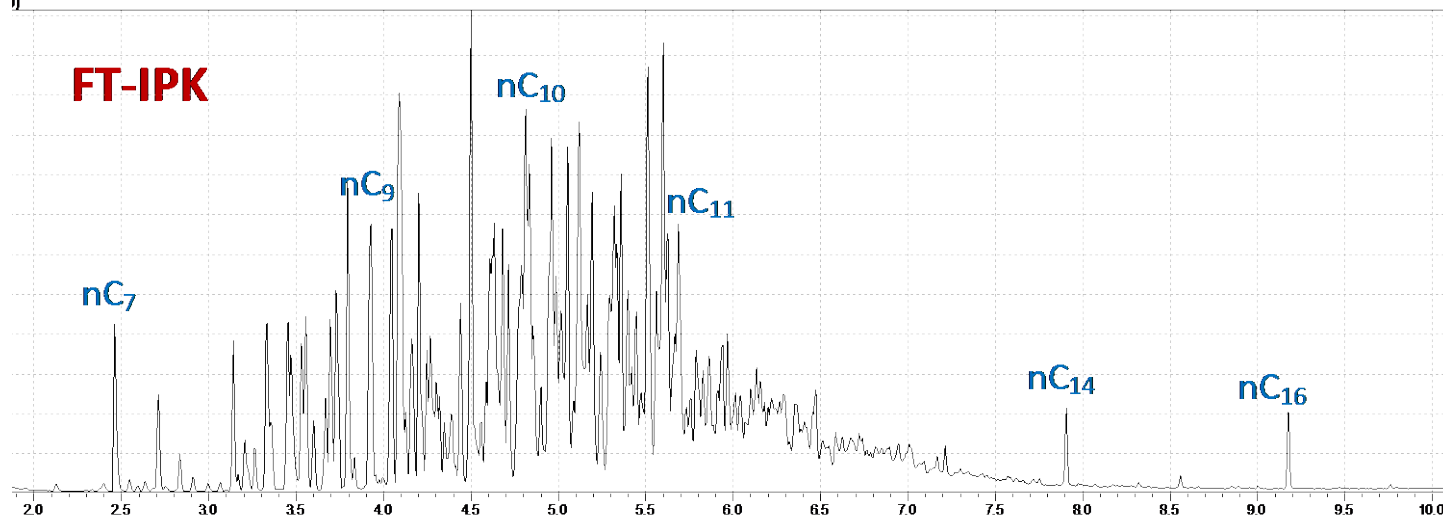
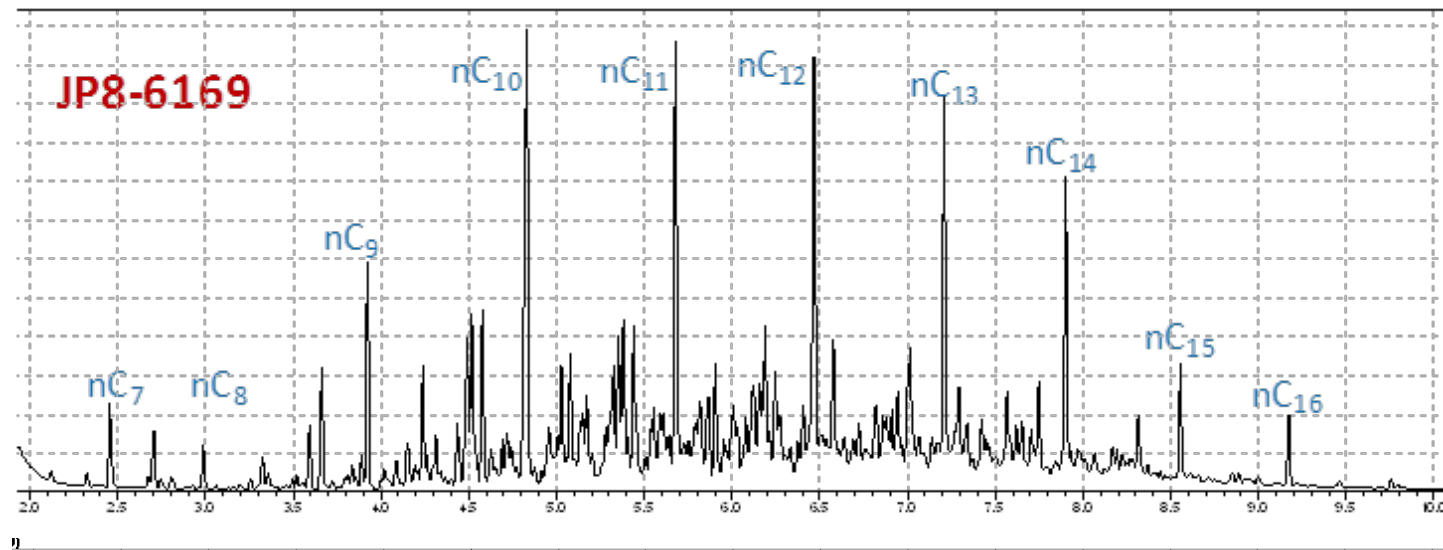
- Current study examined 5 separate fuels:
 - JP-8 6169
 - F-T: Sasol IPK 7629
 - Shell SPK 5729
 - HRJ: Camelina 7720
 - Tallow 6308
- Compositions and combustion characteristics of the fuel properties varied significantly.
- Alternative fuels lack aromatic and cyclo-paraffinic content found in JP-8 but are comprised of larger quantities of iso-paraffins

Fuel data provided by Tim Edwards (AFRL) and Won et al. (AIAA 2013-0156)

GC-MS Chromatography: JP8 vs FT Fuels



GC-MS Chromatography: JP8 vs FT Fuels



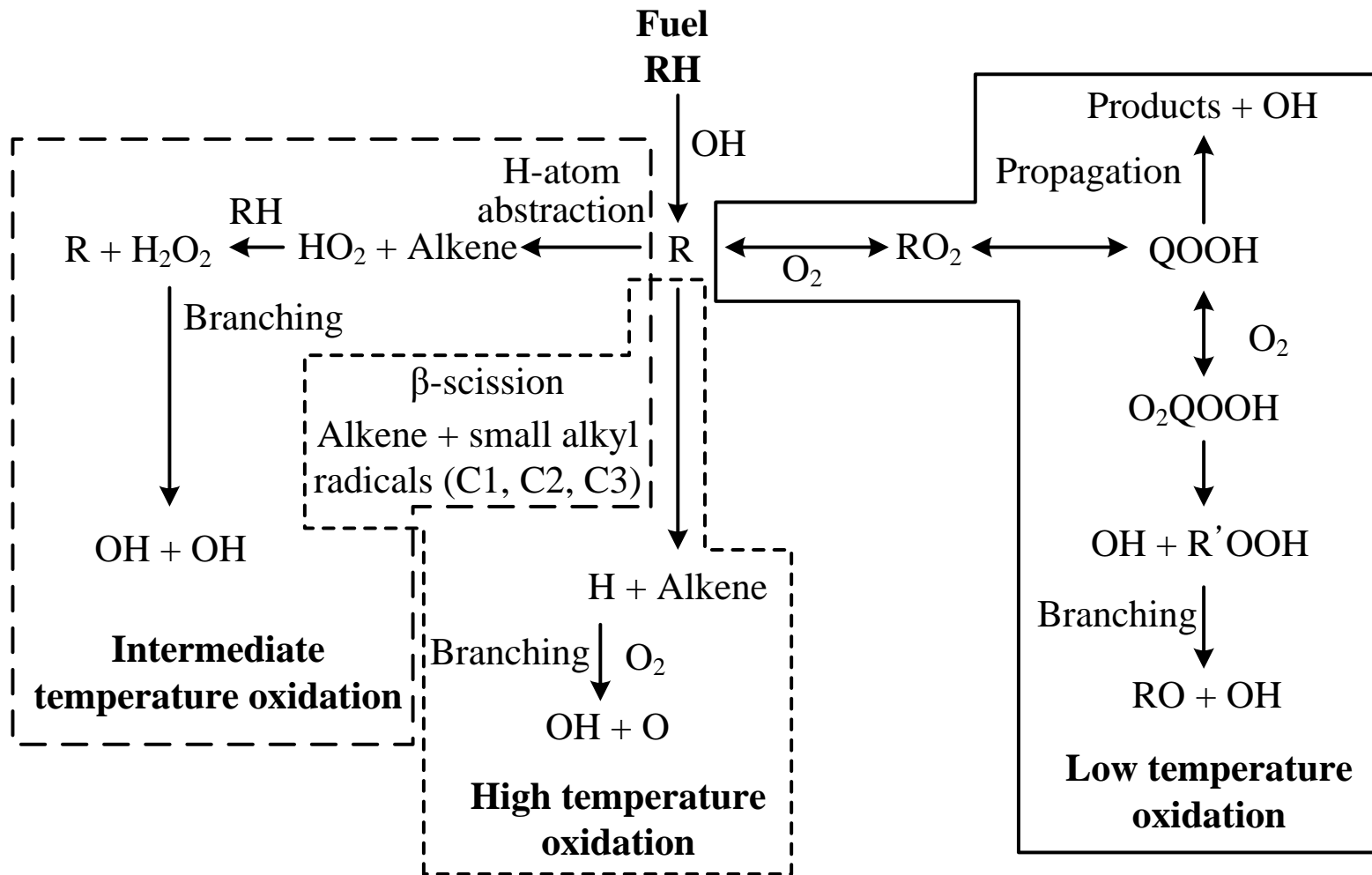
Jet Fuel Surrogate Kinetic Model

- CSE detailed surrogate kinetic model consists of four components to represent chemical groups found in jet fuels
 - n-paraffins: n-Dodecane/n-Decane
 - Iso-paraffins: iso-Octane
 - Aromatics: n-propyl-Benzene
- The surrogate model was validated for a wide range of conditions using kinetic experimental data for ignition delay time, flame speed and species profiles
- The kinetic mechanism also includes detailed nitrogen chemistry for vitiation and NO_x emission predictions
- Surrogate mixture formulation is based on three target criteria:
 - Hydrogen-Carbon Ratio (H/C)
 - Derived Cetane Number (DCN)
 - Threshold Sooting Index (TSI)

Alternative JP-8 Surrogate Mixtures

Surrogate	JP-8 6169	IPK 7629	SPK 5729	Tal 6308	Cam 7720
n-Dodecane	0.382	0.160	0.045	0.595	0.611
n-Decane	0.000	0.000	0.746	0.000	0.000
iso-Octane	0.372	0.647	0.122	0.315	0.292
propyl-Benzene	0.246	0.193	0.087	0.090	0.096
Surrogate Mixture Properties (CSE Calculations)					
DCN	48.1	31.8	58.2	59.0	60.0
TSI	18.25	15.74	9.11	11.09	11.38
H-C Ratio	2.003	2.052	2.134	2.122	2.117
MW	137.1	124.3	138.2	148.1	149.1
Target Fuel Properties (Princeton Measurements)					
DCN	47.3	31.3	58.4	58.1	58.9
TSI	19.29	17.28	9.11	11.58	11.99
H-C Ratio	2.017	2.195	2.237	2.176	2.202
MW	153.9	149.2	136.7	161.0	165.0

Hydrocarbon Oxidation Pathways

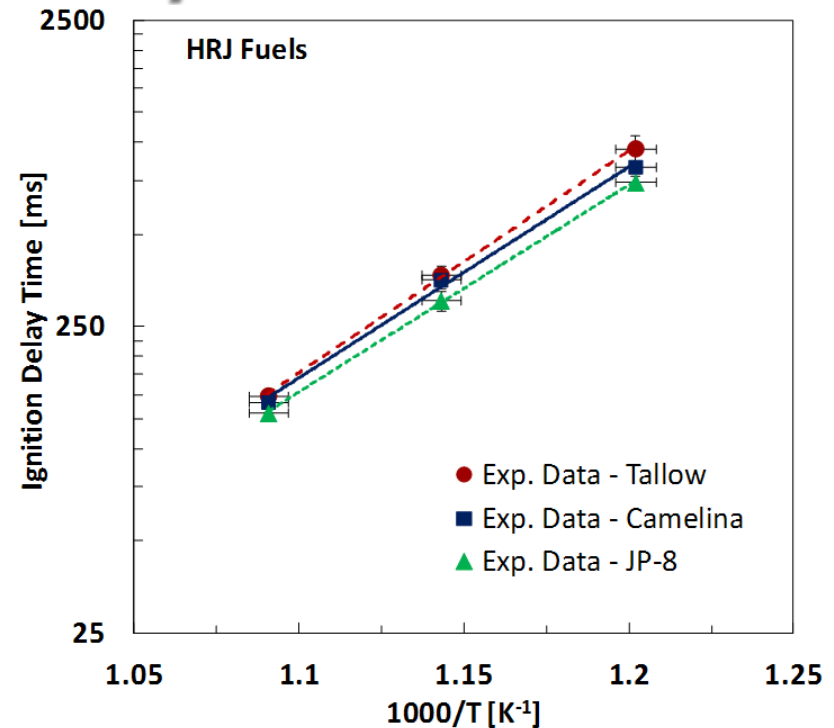
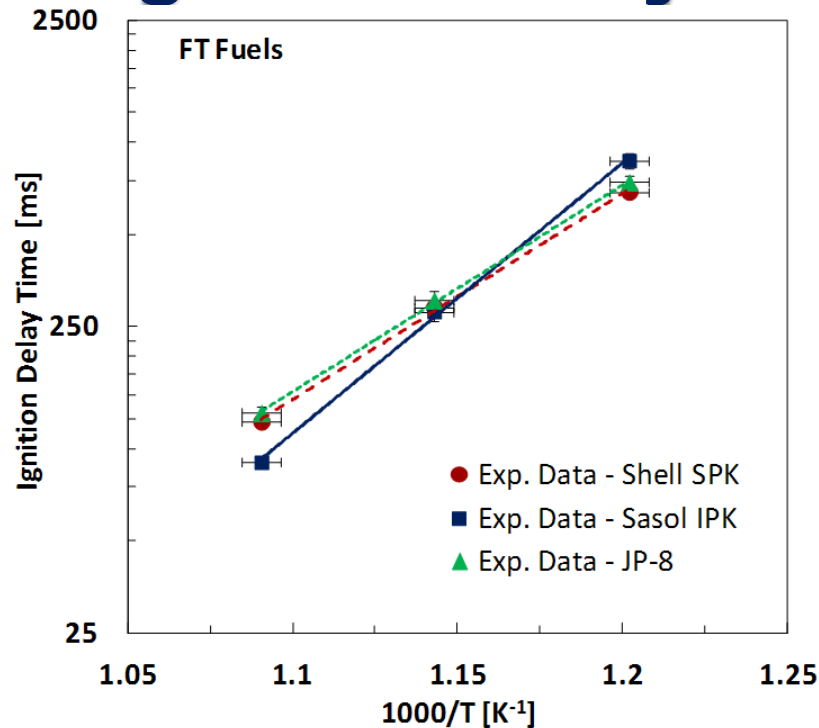


Experimental Conditions and Setup

- Pressure
 - 1 atm
- Temperature
 - 832 K, 875 K, & 917 K
- Equivalence Ratio
 - 0.5, 1.0, & 1.5
- Fuel:
 - JP-8, IPK, SPK, Camelina & Tallow
- Oxidizer Composition:
 - O₂: 17 mol% & 20 mol%
 - NO_x: 0 ppmv – 755 ppmv



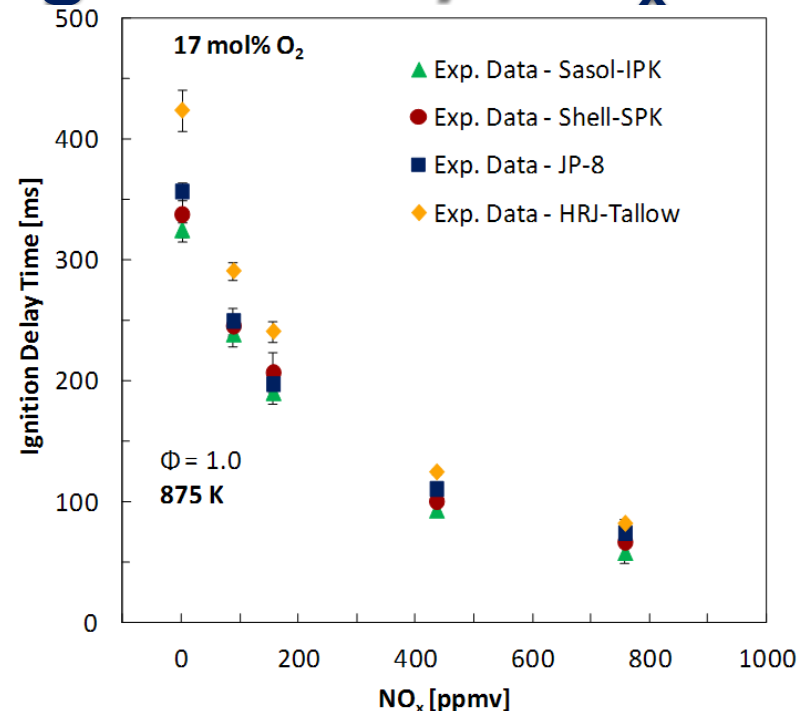
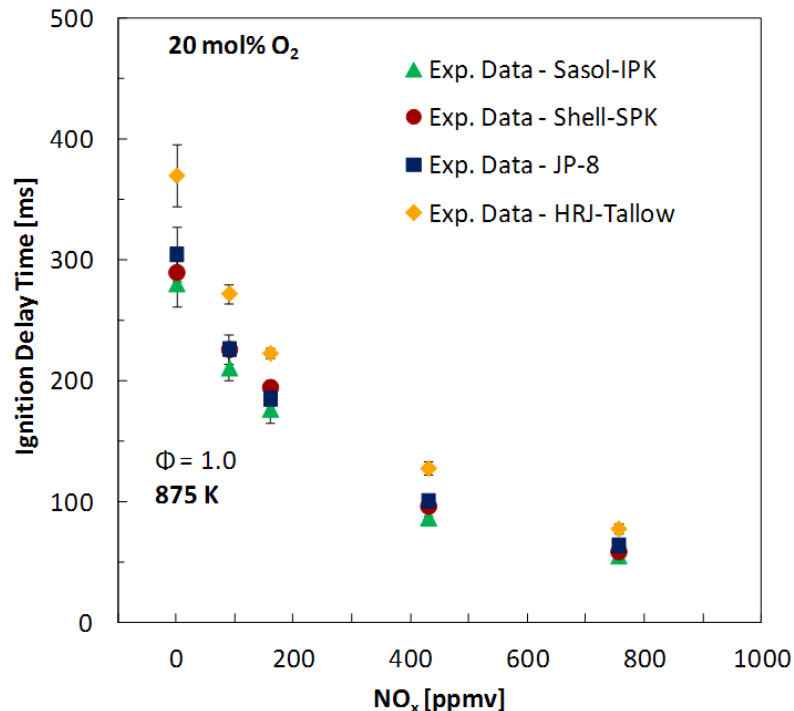
Ignition Delay Time w/out Vitiation



Experimental Activation Energies (832 – 917 K)

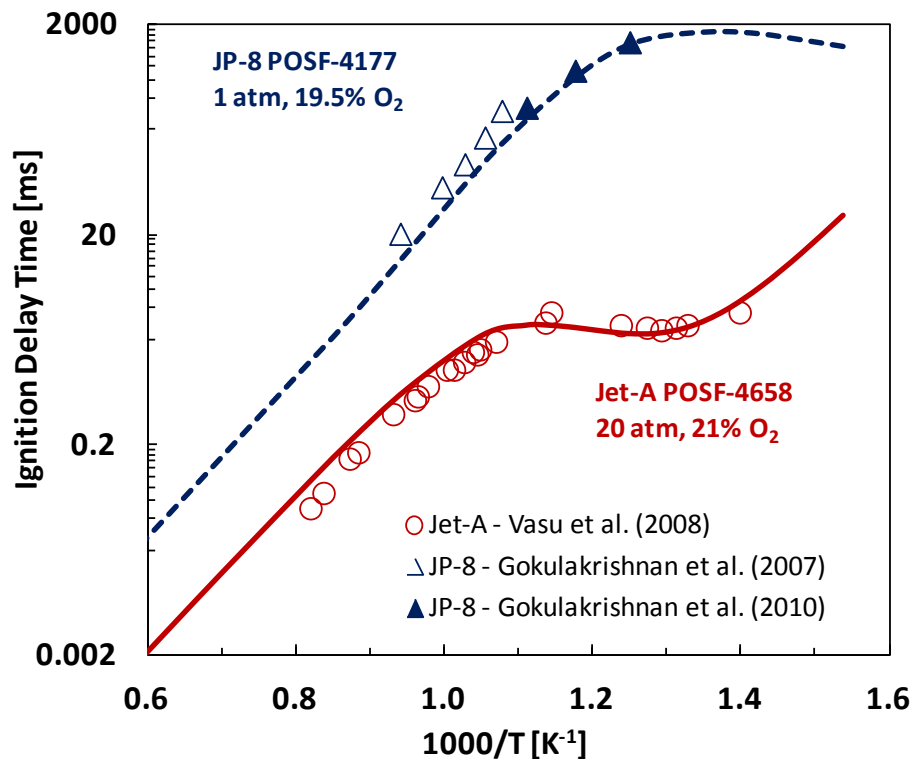
Fuel	E [kcal/mol]	DCN
JP-8	30.90	47.3
Sasol IPK	40.40	31.3
Shell SPK	30.78	58.4
Camelina	31.49	58.9
Tallow	33.11	58.1

Alternative Fuel Ignition w/ NO_x



- Variation in IDT between the fuels reduces with increased NO_x
- Sasol IPK has shorter IDT despite having the lowest DCN
- HRJ Tallow has longer IDT while DCN is higher than JP-8 and Sasol IPK
- The marginal effect on IDT diminishes with an asymptotic behavior as NO_x concentration approaches 800 ppm

Surrogate Model Development



- CSE jet fuel surrogate kinetic model is a four-component surrogate kinetic model validated for each individual component.
- n-Paraffins:
 - n-decane & n-dodecane
- iso-Paraffins:
 - iso-octane
- Aromatics:
 - n-propyl-benzene
- For Jet-A and JP-8 mixtures the surrogate model predicts ignition delay time reasonably well but requires development for alternative fuels and vitiated combustion

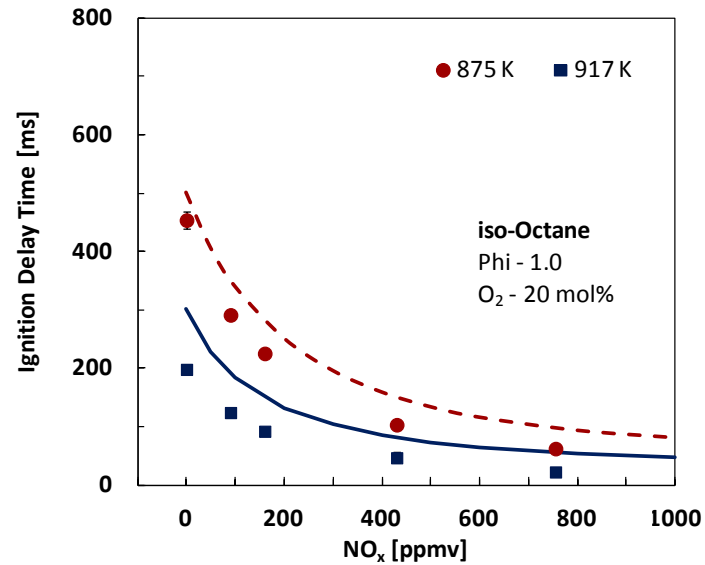
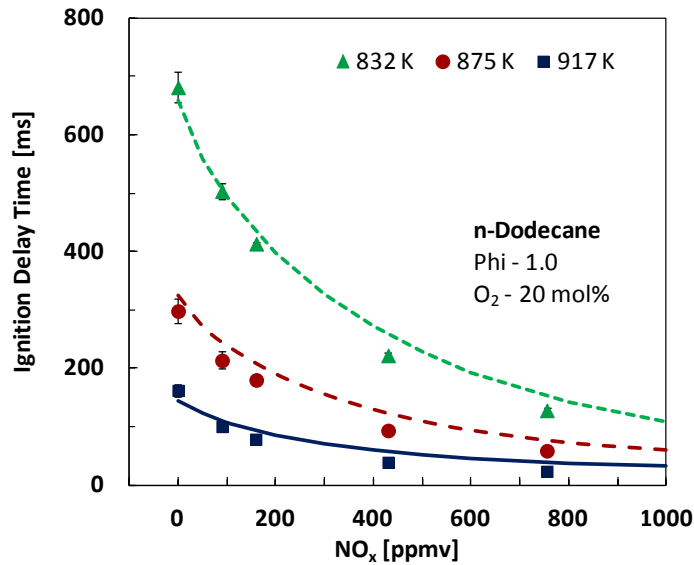
Surrogate Model Composition

- Surrogate mixture compositions are determined by matching the following target characteristics (Dooley et al., 2010) :
 - hydrogen-to-carbon ratio (H/C)
 - derived cetane number (DCN)
 - threshold sooting index (TSI)
 - mean molecular weight (MW)
- Surrogate mixture compositions (mol %) for the fuels examined in this study are provided below:

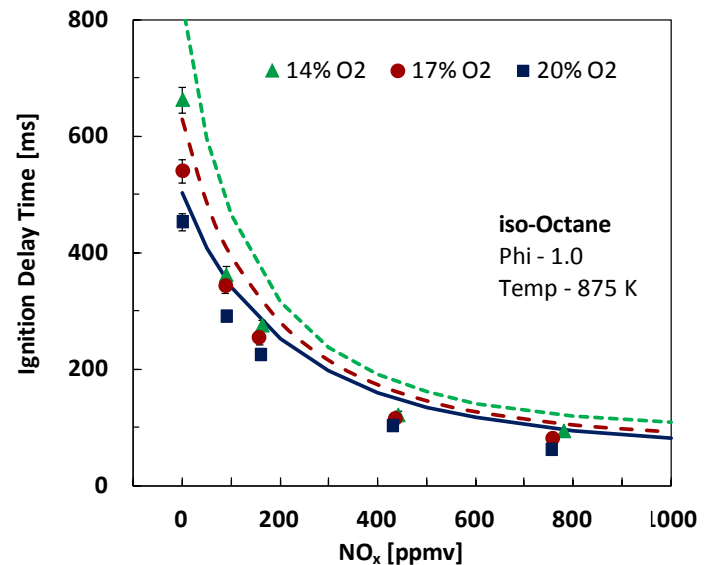
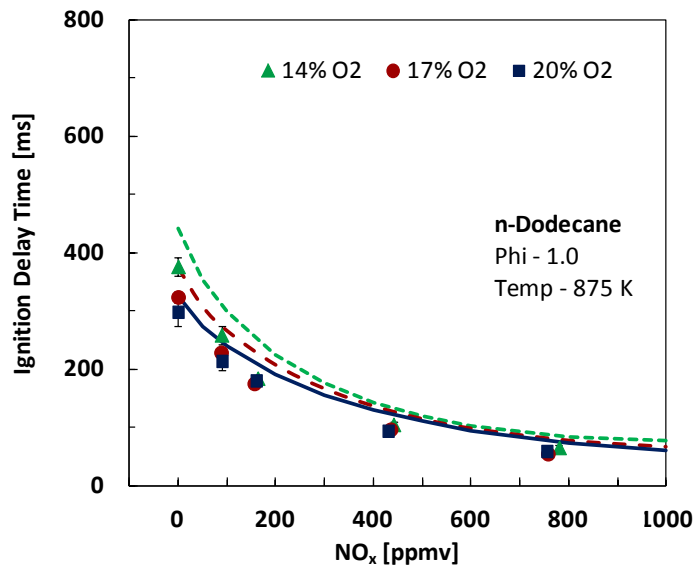
Surrogate	JP-8	Sasol IPK	Shell SPK	HRJ Camel.	HRJ Tallow
n-decane	0.0	0.0	74.6	0.0	0.0
n-dodecane	38.2	16.0	4.5	61.1	59.5
iso-octane	37.2	64.7	12.2	29.2	31.5
propyl-benzene	24.6	19.3	8.7	9.6	9.0

Surrogate Component Experiments

Effect of Temperature

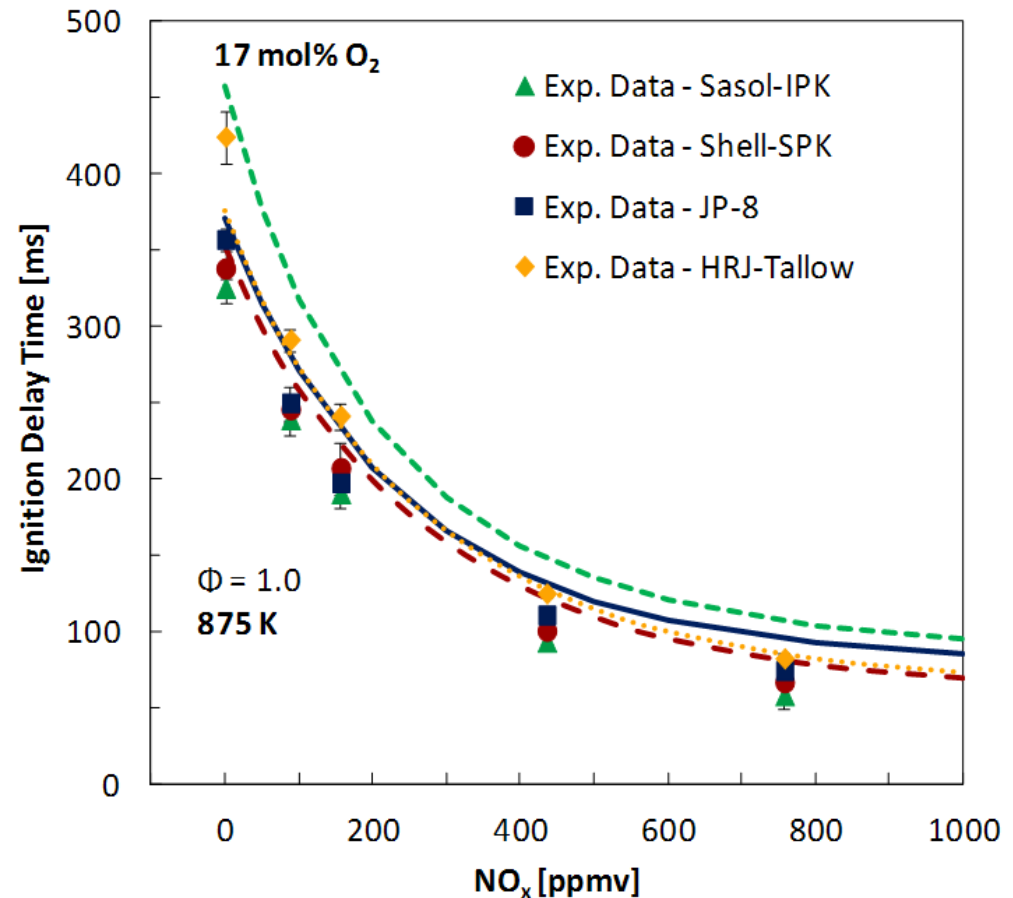


Effect of Oxidizer O_2

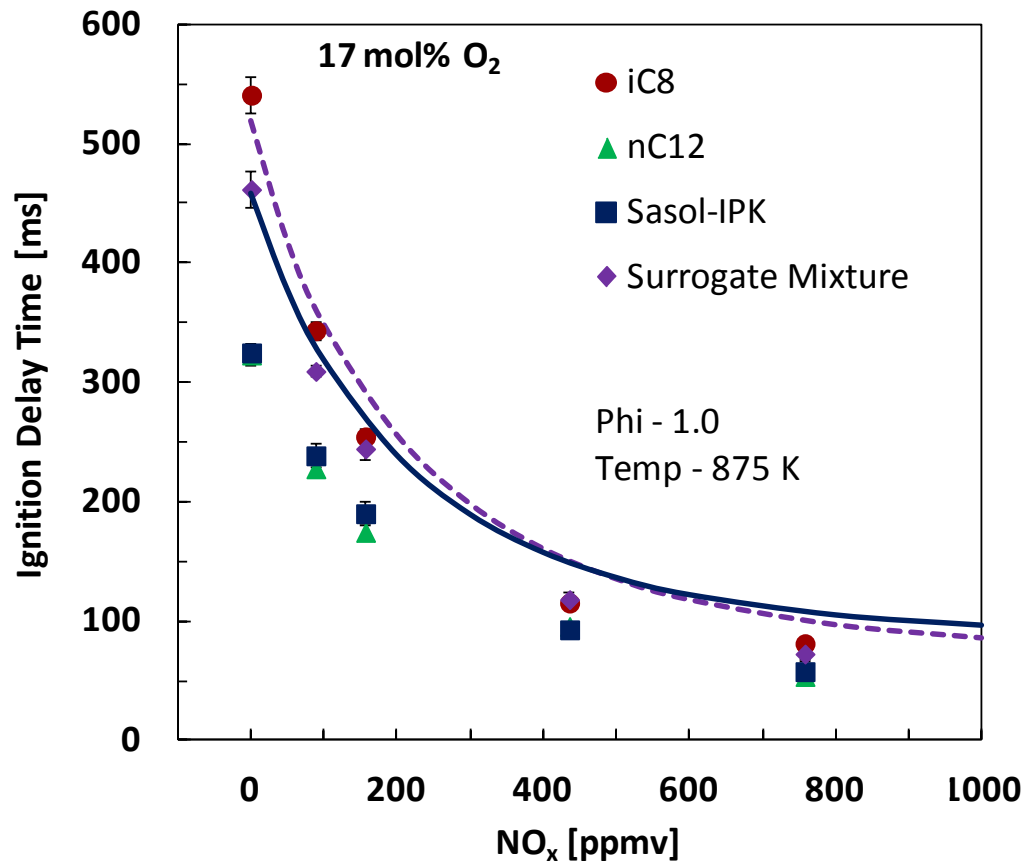


Surrogate Model Comparison

- Overall the model predicts effect of NO_x on IDT for JP-8, SPK and HRJ Tallow reasonably well.
- The model does not predict IDT well for the Sasol IPK fuel.
- Sasol IPK: ~95% iso-paraffins with the lowest DCN
- Current model uses highly-branched iso-octane to model iso-paraffinic fuel components



Sasol IPK Surrogate Mixtures



- 2-component surrogate for IPK, comprised of 82 mol % iso-octane and 18 mol % n-dodecane (DCN = 31) , was tested in the flow reactor
- Model predicts IDT of the iC₈/nC₁₂ mixture significantly better than the it does for the IPK fuel
- Work in progress to investigate weakly-branched iso-paraffinic fuels to improve the surrogate model

Modeling of Jet Fuel-NO_x Kinetics

- NO_x promotes oxidation in the low- and intermediate-temperature regimes through fuel independent and fuel dependent pathways
- Major fuel-independent reaction pathways of NO_x on jet fuel oxidation promotes OH radical production:
 - $\text{HO}_2 + \text{NO} = \text{NO}_2 + \text{OH}$
- Fuel dependent reactions involve NO₂ to form alkyl radicals and HONO/HNO₂ and eventually NO and OH:
 - $\text{C}_x\text{H}_y + \text{NO}_2 = \text{C}_x\text{H}_{y-1} + \text{HONO/HNO}_2$
 - $\text{HNO}_2 + \text{M} = \text{HONO} + \text{M}$
 - $\text{HONO} + \text{M} = \text{NO} + \text{OH} + \text{M}$
- Fuel dependent HC-NO_x interaction is relatively better understood for natural gas surrogate fuels (Gokulakrishnan et al. CNF 2014) than gasoline or jet fuels.

Lean, Premixed Prevaporized Combustion

- Ground-based power production is highly regulated for pollution emissions (NO_x, CO, particulates)
 - Lean, Premixed combustion allows for control of flame temperature
 - Lower flame temperature reduces NO_x production and minimizes soot production
- Typical Lean, Premixed Gas Turbines burn natural gas
- Many locations do not have access to low-cost natural gas
 - Need pipeline infrastructure to move natural gas from wells to users
 - Liquified natural gas (LNG) requires significant processing , adding expense
 - Liquid fuels typically have higher energy density and are easier to transport
- Techniques for 'clean' combustion of liquid fuels are needed

LPP Combustion Concept

LPP = Lean, Premixed & Prevaporized

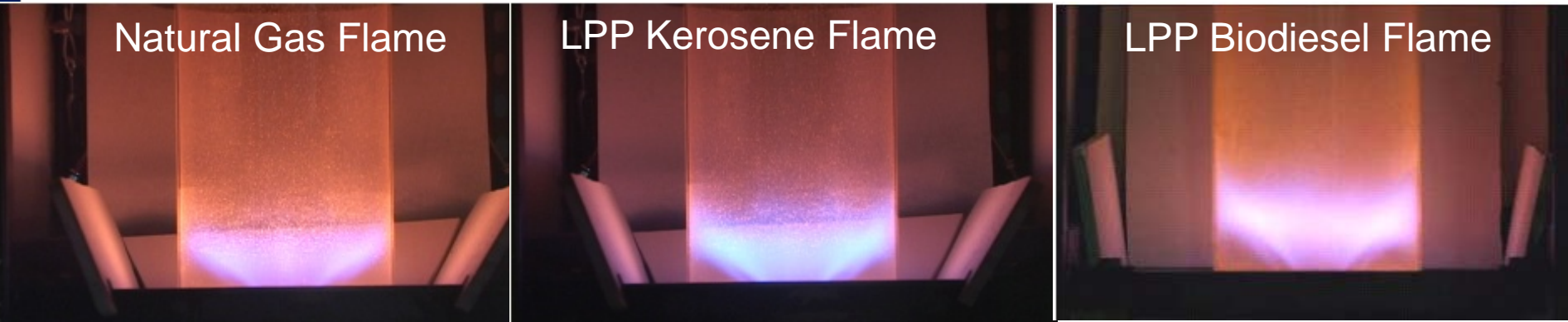
- LPP Skid converts liquid fuel into a substitute natural gas (LPP Gas) by vaporizing the liquid fuel into a reduced-oxygen background gas (diluent).
- LPP Gas properties adjusted with choice/quantity of diluent to match natural gas characteristics (heating value, Modified Wobbe Index, etc.)
- LPP Gas is burned with low emissions in place of natural gas in combustion device.
- Burning renewable liquid fuels (biodiesel, ethanol, etc.) creates a low-emissions, renewable energy power plant with no net CO₂ emissions.

Combustor Technology Review

Traditional Combustion of **Liquid Fuels** in a Spray (Diffusion) Flame Creates High Levels of NO_x, CO and Particulate Matter, even with Significant Water Injection to Reduce Emissions.

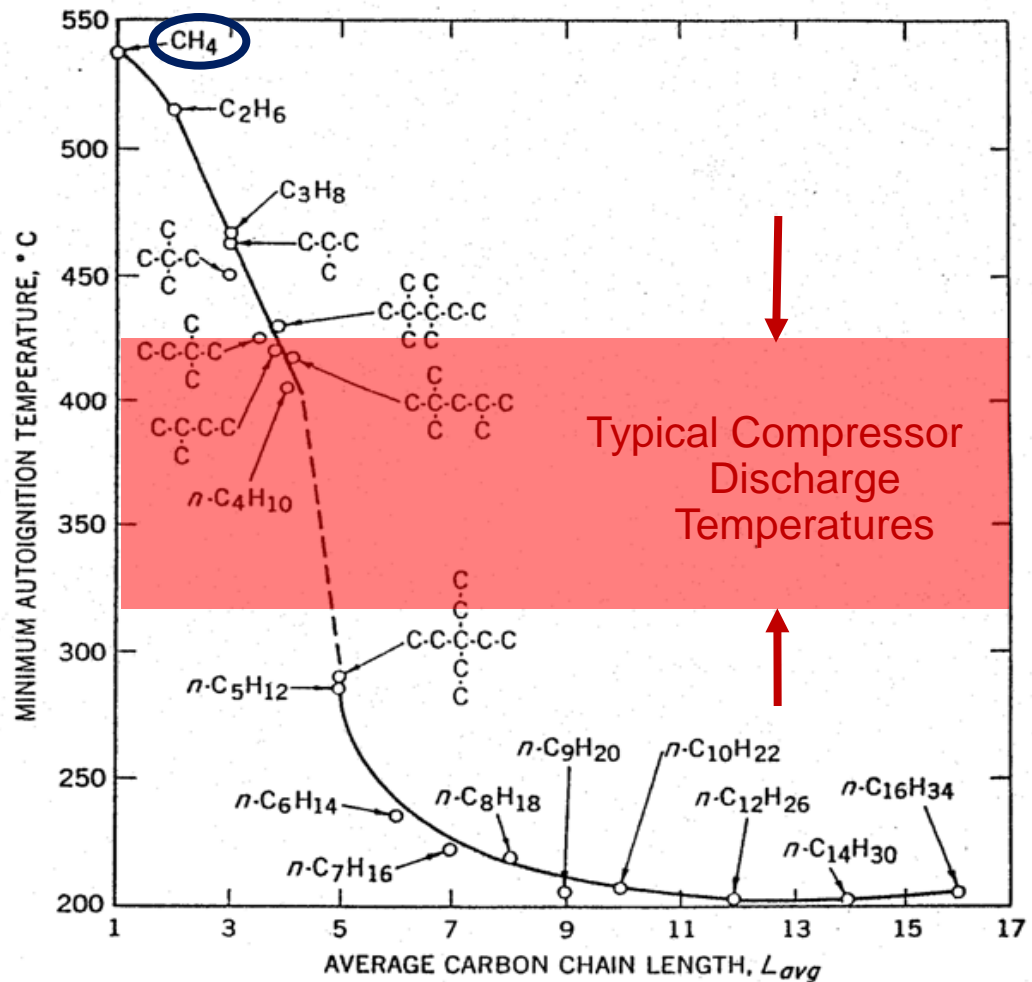


Gaseous Fuel Combustion (with **Natural Gas** or LPP Gas) in a Lean, Premixed Burner Creates a Low-Emissions Combustion

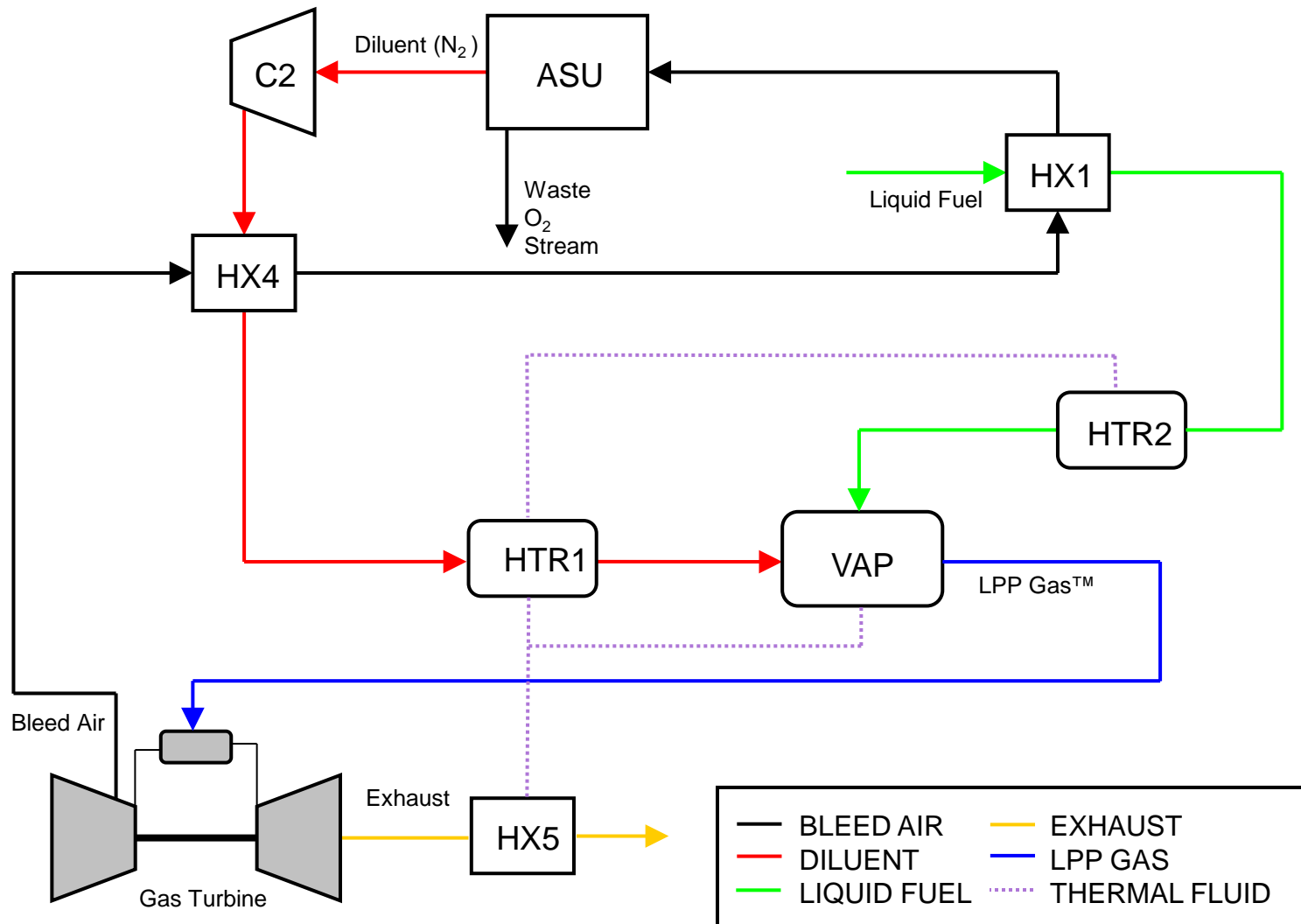


Fuel Autoignition Characteristics

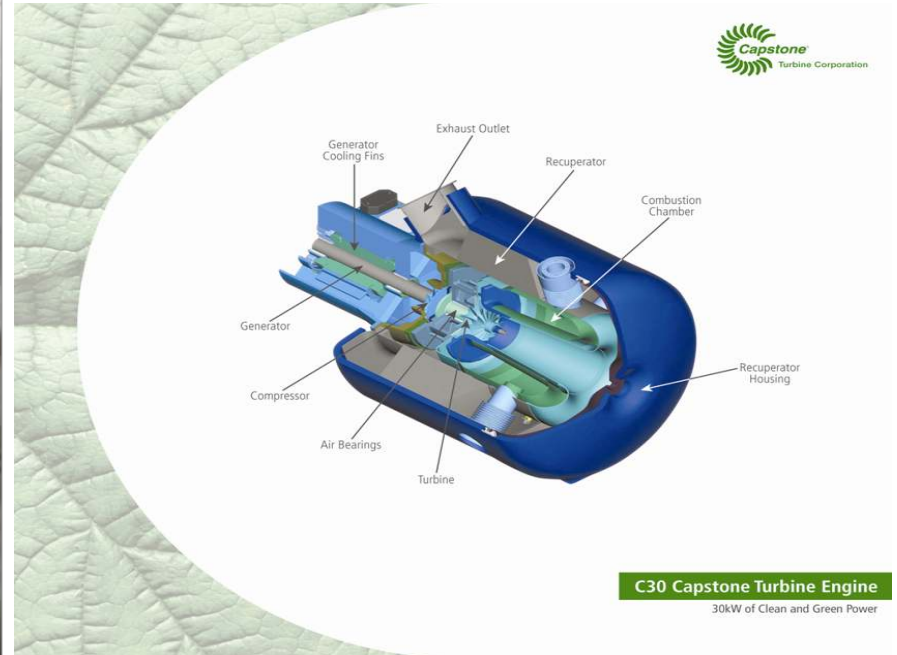
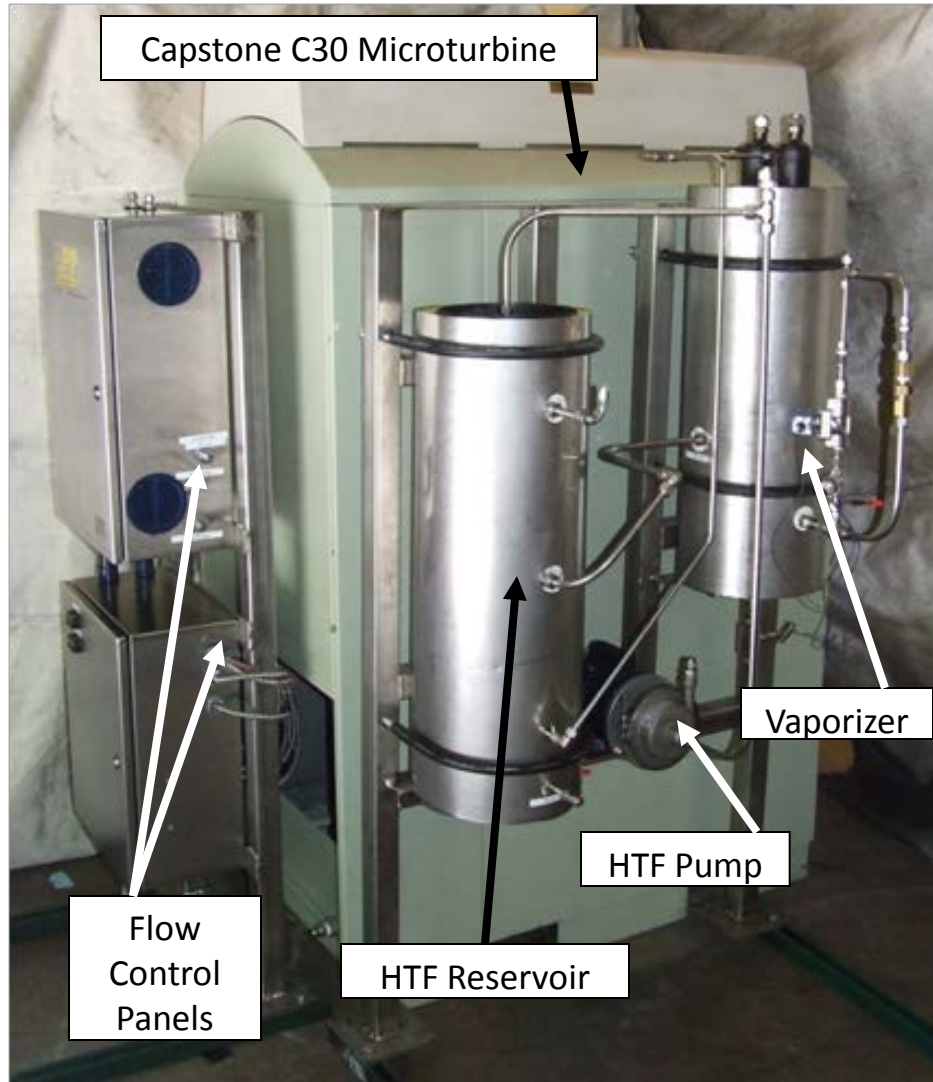
Autoignition becomes a problem for **higher hydrocarbons**, at higher inlet temperatures, where it is not a problem for **natural gas**



Gas Turbine/LPP Process Flow Diagram

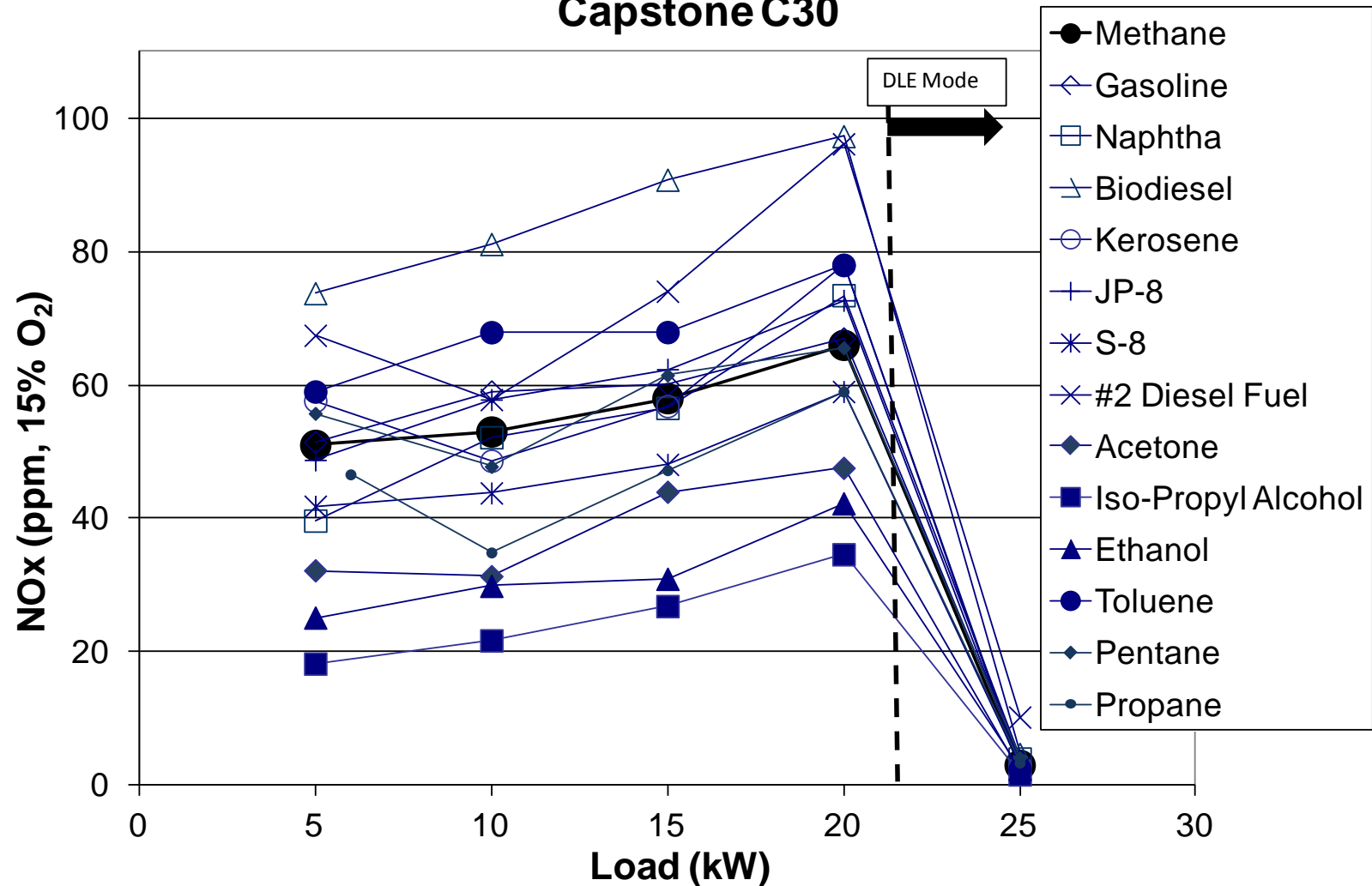


Demonstration Unit

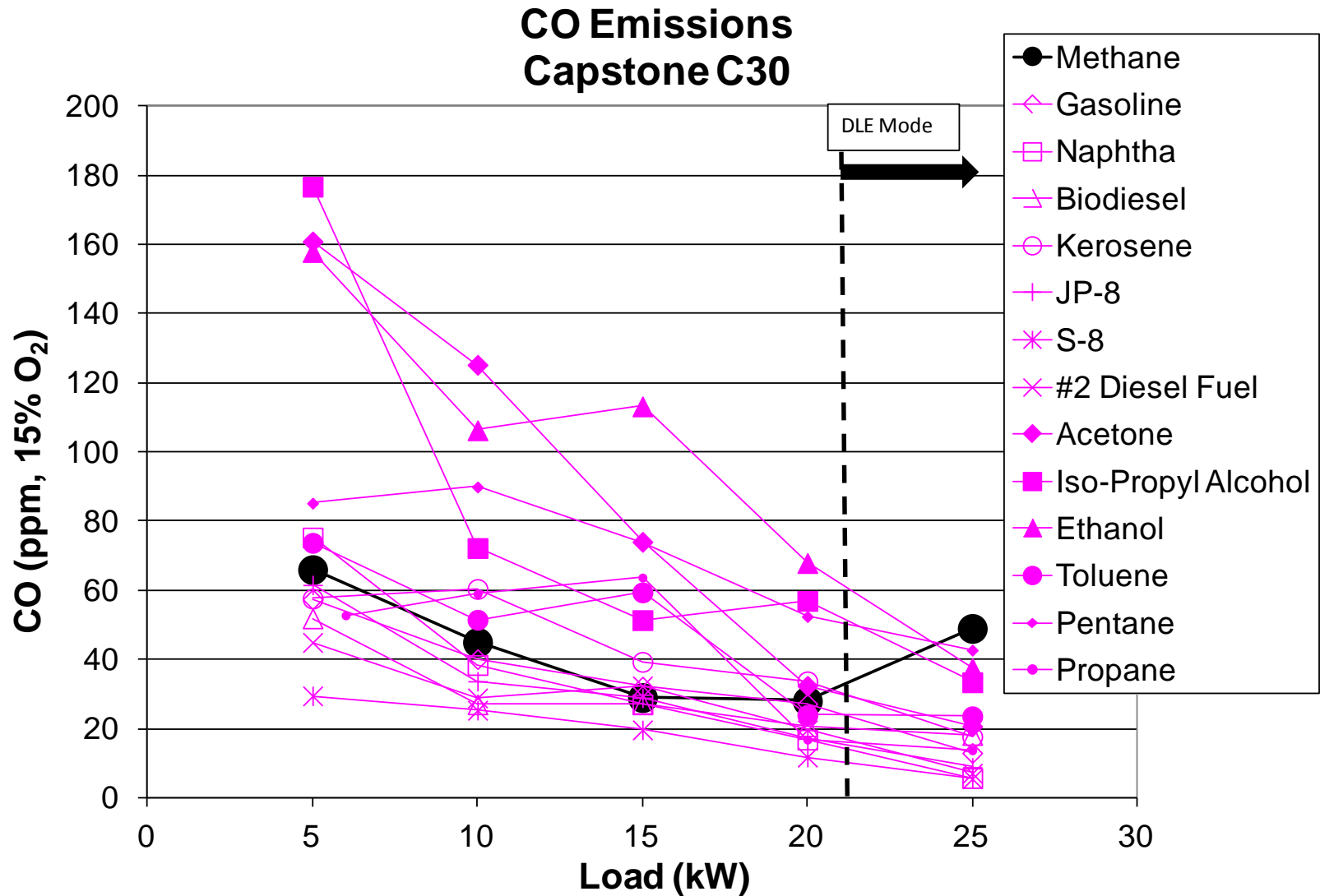


Premixed Combustion of Liquid Fuels

NO_x Emissions Capstone C30



Premixed Combustion of Liquid Fuels



Conclusions

- Alternative fuels are viable 'drop-in' replacements for jet engines
- Differences in chemical composition can alter the combustion behavior of these fuels under both standard and vitiated conditions
- Current 4-component surrogate model predicts IDT for JP-8 and certain alternative fuels at the experimental conditions of this study with reasonable accuracy
- Vitiating combustion aspects of the model, specifically NO_x chemistry, are found to predict the enhancement of ignition, accounting for the change of reaction pathways in the low- and intermediate-temperature regimes
- Lean, Premixed, Prevaporized combustion is a viable option for burning alternative liquid fuels in gas turbines
 - Emissions are similar to those found for natural gas combustion
 - Wide-range of liquid fuels have been tested

Future Work

- Future research will focus on improvement of the surrogate model at both standard air and vitiated conditions including:
 - Inclusion of additional surrogate component(s) to represent larger, less-branched iso-paraffins
 - Investigation of additional target criteria to improve surrogate formulation

Acknowledgements

- U.S. Air Force Research Laboratory
- Dr. Barry Kiel, Air Force Research Laboratory
- Bethany Huelskamp at ISSI
- Dr. Tim Edwards, Air Force Research Laboratory

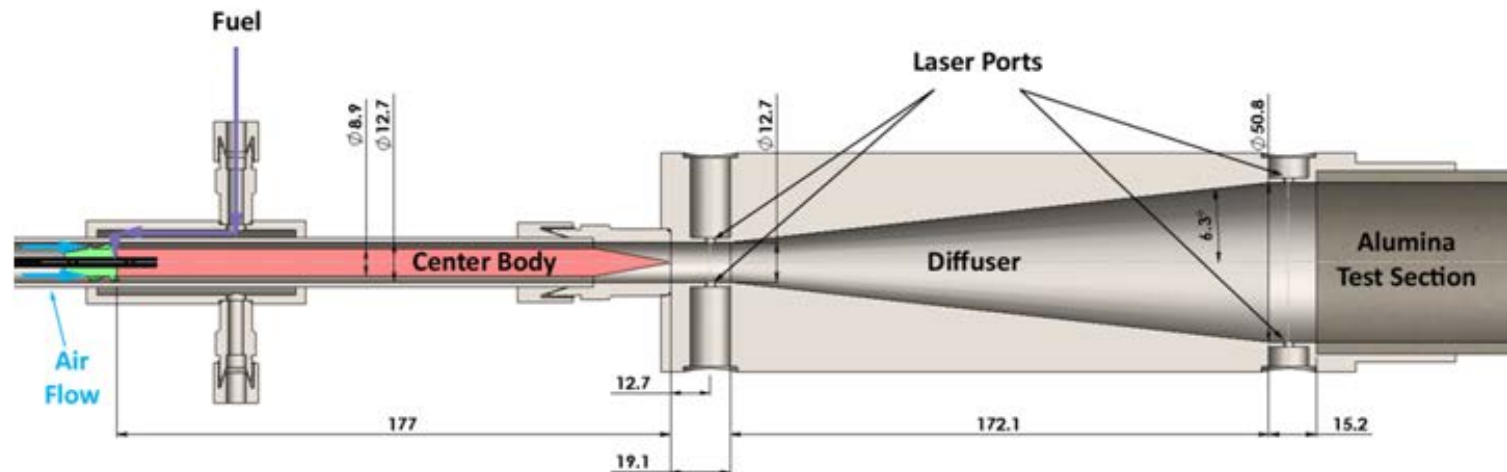
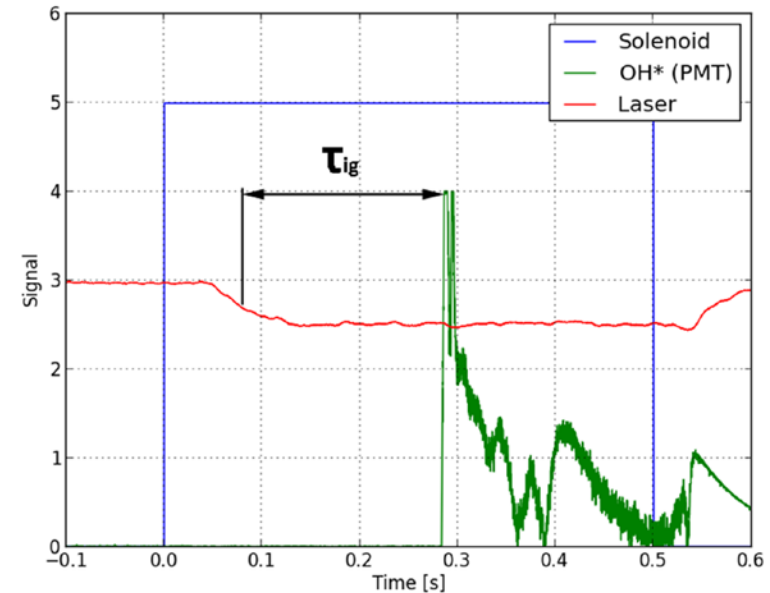
Thank You

- Questions?
- Contact:
 - mklassen@csefire.com

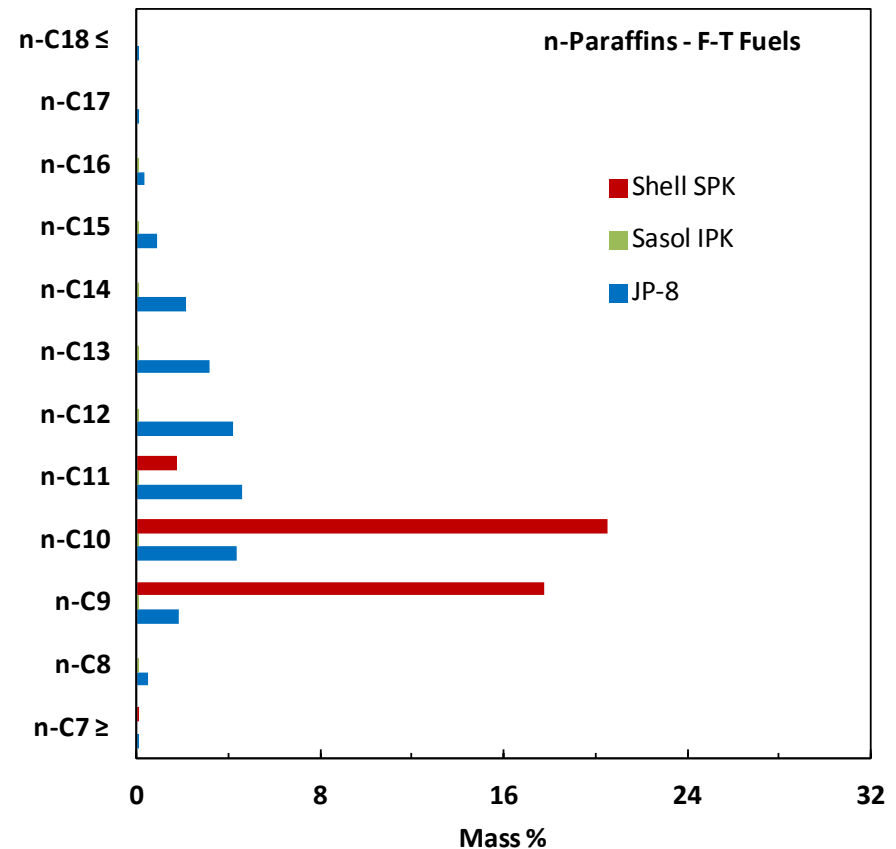
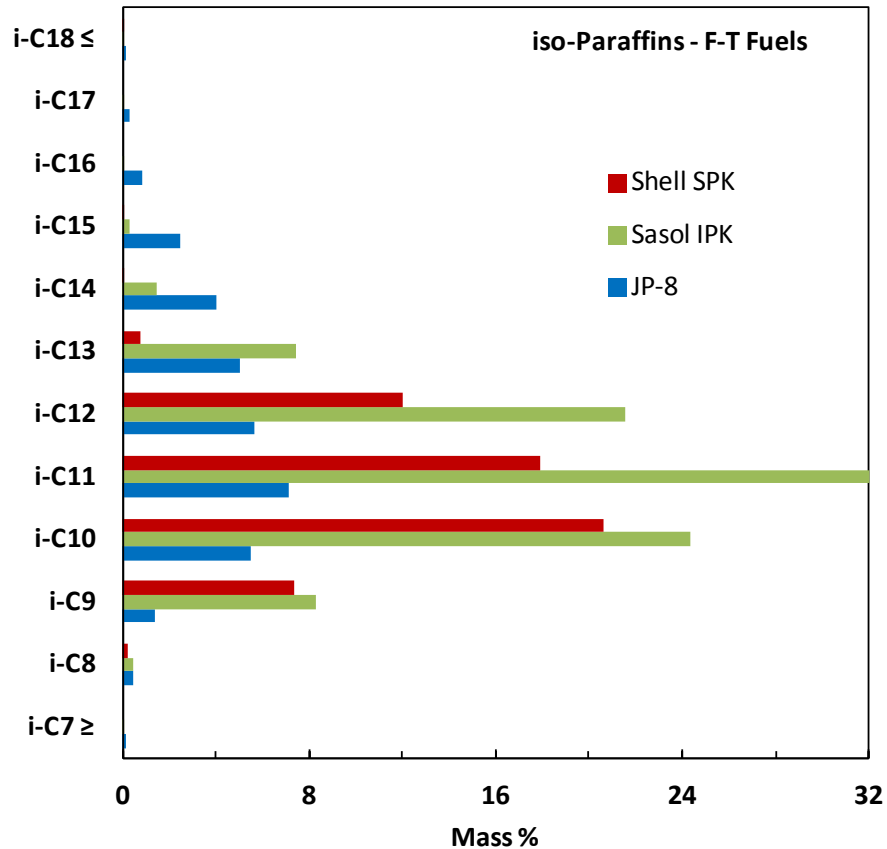
Extra Slides

Ignition Delay Time Measurement

- Absorption of infrared HeNe beam ($3.39\ \mu\text{m}$) by H-C bonds in the fuel/oxidizer mixture used to determine time delay between solenoid activation and fuel injection into the reactor
- OH^* excitation measured by PMT equipped with $310 (\pm 5)\ \text{nm}$ band pass filter to determine ignition event
- IDT (τ_{ig}) measured as the time difference between initial PMT excitation and 50% reduction in the HeNe laser signal at the diffuser entrance

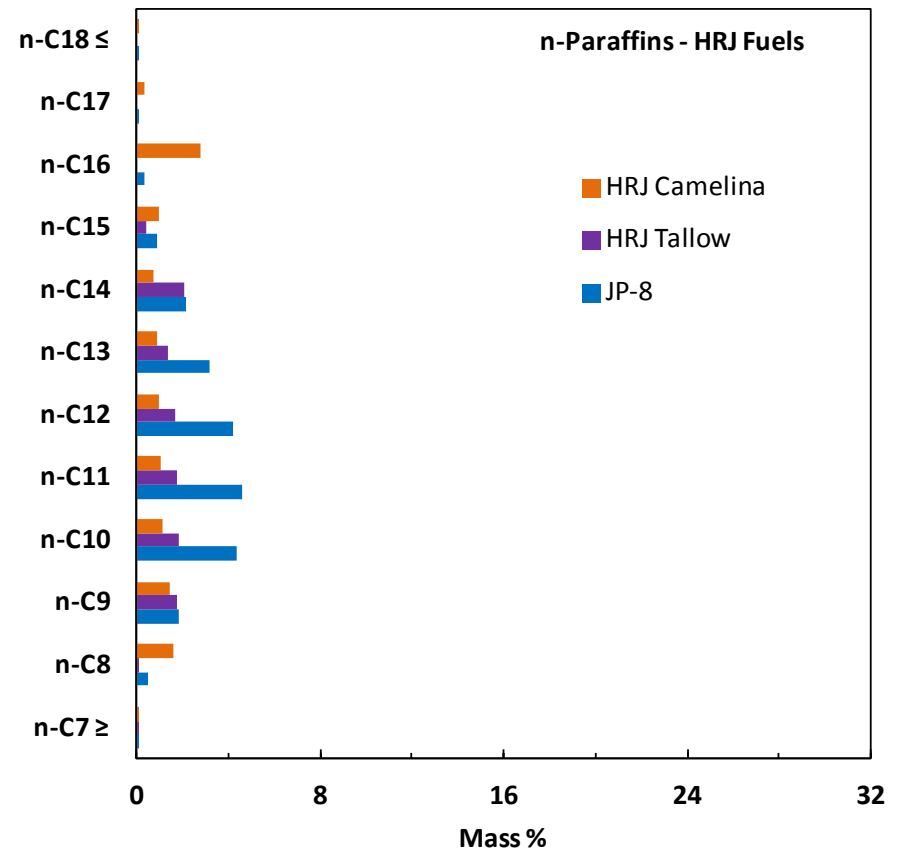
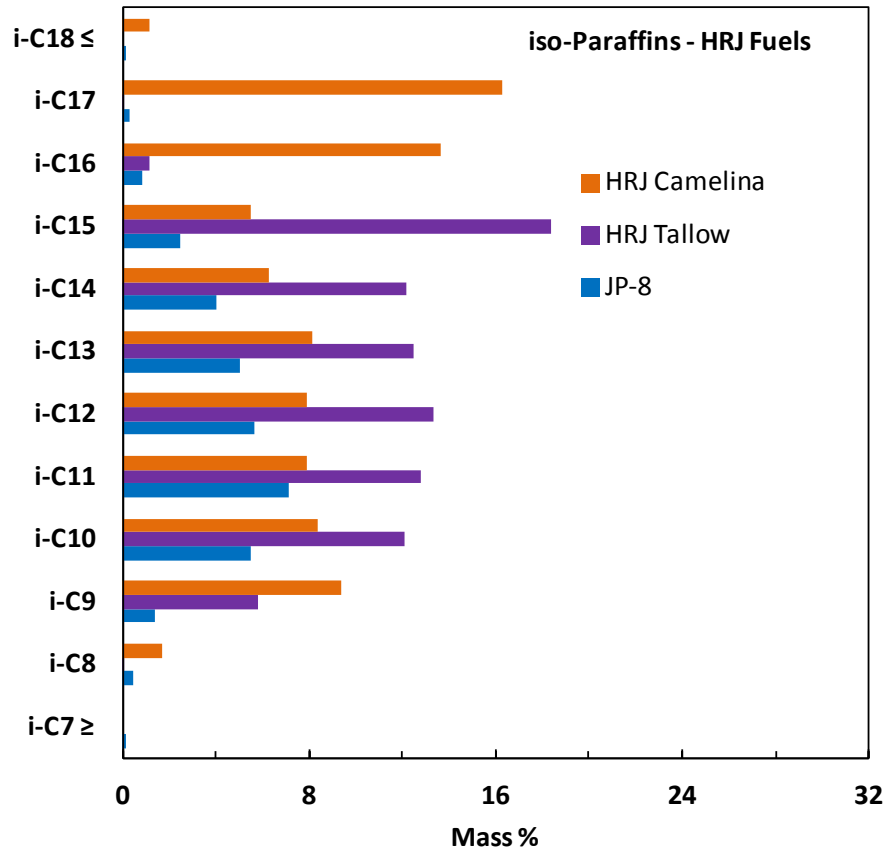


JP-8 & Fischer-Tropsch Fuel Comparisons



Data provided by Tim Edwards (AFRL) 2014

JP-8 & Hydrotreated Fuel Comparisons



Data provided by Tim Edwards (AFRL) 2014

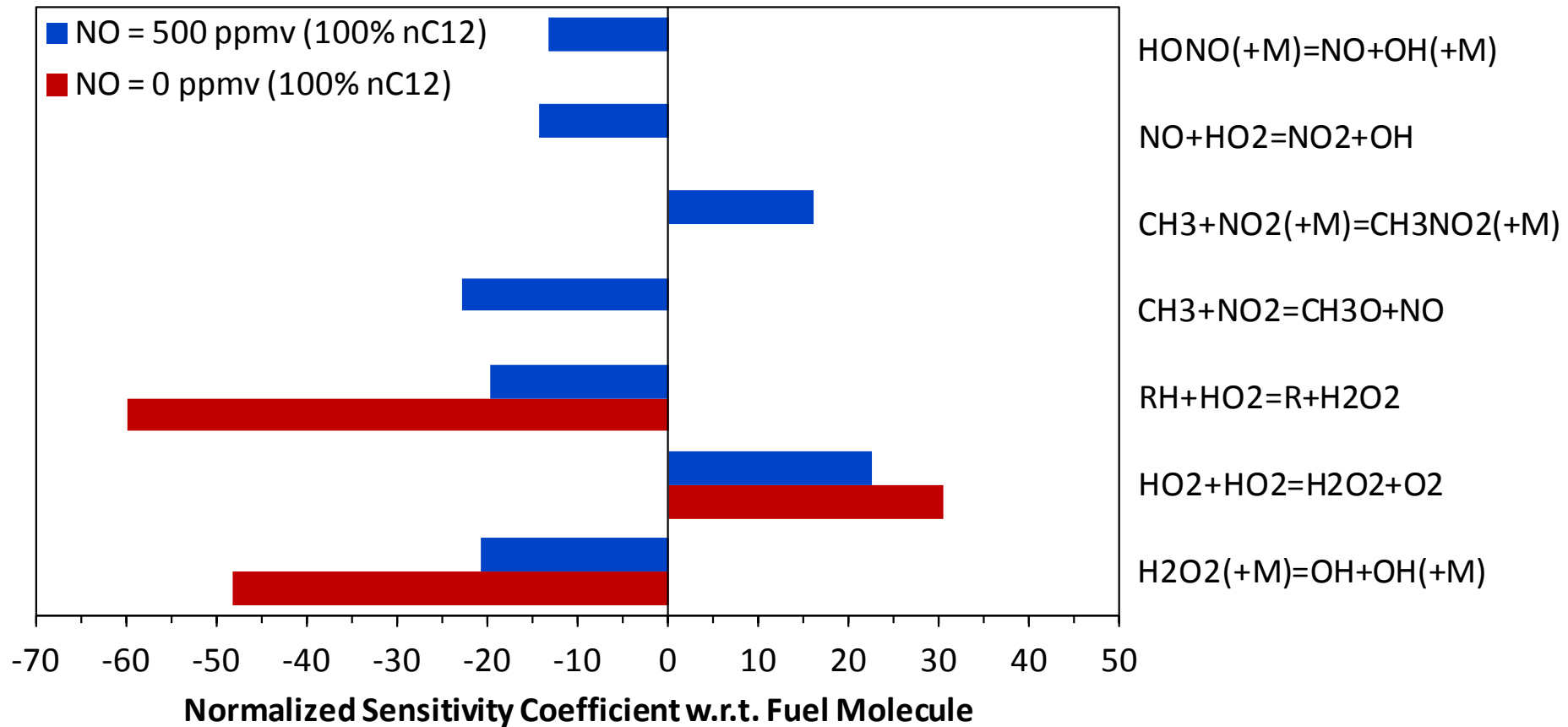
Jet Fuel Comparisons

- Five separate jet fuels were examined in this study:
 - Petroleum based: JP-8 6169
 - Fischer-Tropsch Fuels: Sasol IPK 7629 & Shell SPK 5729
 - Hydrotreated Renewable Jet Fuels: Camelina 7720 & Tallow 6308

Jet Fuel Type	Air Force POSF #	Density ^a [kg/m ³]	DCN ^b	TSI ^b	H/C Ratio ^b	MW ^b [g/mol]	Molecular Formula ^b
JP-8	6169	785	47.3	19.28	2.02	153.9	C _{11.0} H _{22.1}
F-T Sasol IPK	7629	739	31.3	17.28	2.20	149.2	C _{10.5} H _{23.0}
F-T Shell SPK	5729	730	58.4	9.11	2.24	136.7	C _{9.6} H _{21.4}
HRJ Camelina	7720	752	58.9	11.99	2.20	165.0	C _{11.6} H _{25.5}
HRJ Tallow	6308	748	58.1	11.58	2.18	161.0	C _{11.3} H _{24.7}

^a – Edwards (2014); ^b – Won et al. (2013)

Reaction Sensitivity for NO_x Addition



Jet Fuel Chemical Class Composition

